

Table 4.1 The effects of the disk rotation on the critical condition for appearance of the inertia-driven vortex roll ($\Delta T=0$, $H=20.0$ mm and $D_j=10.0$ mm).

| Vortex Roll | Disk rotation speed (Ω , rpm) | Flowrate (Q_j , SLPM) | Re_Ω | Re_j |
|-------------------------------|---------------------------------------|--------------------------|-------------|--------|
| Primary inertia-driven roll | 0 | 0.1 | 0 | 14 |
| | 5 | 0.2 | 389 | 27 |
| | 10 | 0.3 | 778 | 41 |
| | 15 | 0.4 | 1,168 | 54 |
| Secondary inertia-driven roll | 0 | 1.3 | 0 | 180 |
| | 5 | 1.4 | 389 | 190 |
| | 10 | 1.6 | 778 | 216 |
| | 15 | 1.9 | 1,168 | 257 |
| Tertiary inertia-driven roll | 0 | 5.0 | 0 | 676 |
| | 10 | 5.0 | 778 | 676 |
| | 15 | 5.3 | 1,168 | 717 |
| | 20 | 5.8 | 1,557 | 784 |

Table 4.2 The effects of the disk rotation on the critical condition for onset of the buoyancy-driven vortex flow (H=20.0 mm and D_j=10.0 mm).

| Disk rotation speed (Ω , rpm) | Flowrate (Q _j , SLPM) | Temperature difference ΔT (°C) | Re _j | Gr | Ra | Gr/Re _j ² |
|---|-------------------------------------|--|-----------------|--------|--------|---------------------------------|
| Re _Ω = 0 (Ω = 0 rpm) | 5.5 | 2.2 | 744 | 2,362 | 1,654 | 0.0043 |
| | 7.2 | 4.0 | 974 | 4,295 | 3,007 | 0.0045 |
| | 9.4 | 7.0 | 1,271 | 7,516 | 5,621 | 0.0046 |
| | 11.4 | 10.0 | 1,542 | 10,737 | 7,516 | 0.0045 |
| Re _Ω = 389 (Ω = 5 rpm) | 5.5 | 6.4 | 744 | 6,872 | 4,810 | 0.0118 |
| | 7.2 | 10.1 | 974 | 10,845 | 7,590 | 0.0115 |
| | 9.4 | 17.4 | 1,271 | 13,530 | 9,470 | 0.0116 |
| | 11.4 | 25.3 | 1,542 | 27,165 | 19,016 | 0.0114 |
| Re _Ω = 778 (Ω = 10 rpm) | 5.5 | 12.1 | 744 | 12,990 | 9,095 | 0.0235 |
| | 7.2 | 20.5 | 974 | 22,012 | 15,408 | 0.0232 |
| | 9.4 | 34.8 | 1,271 | 37,370 | 26,160 | 0.0232 |
| | 11.4 | 51.6 | 1,542 | 55,405 | 38,785 | 0.0233 |

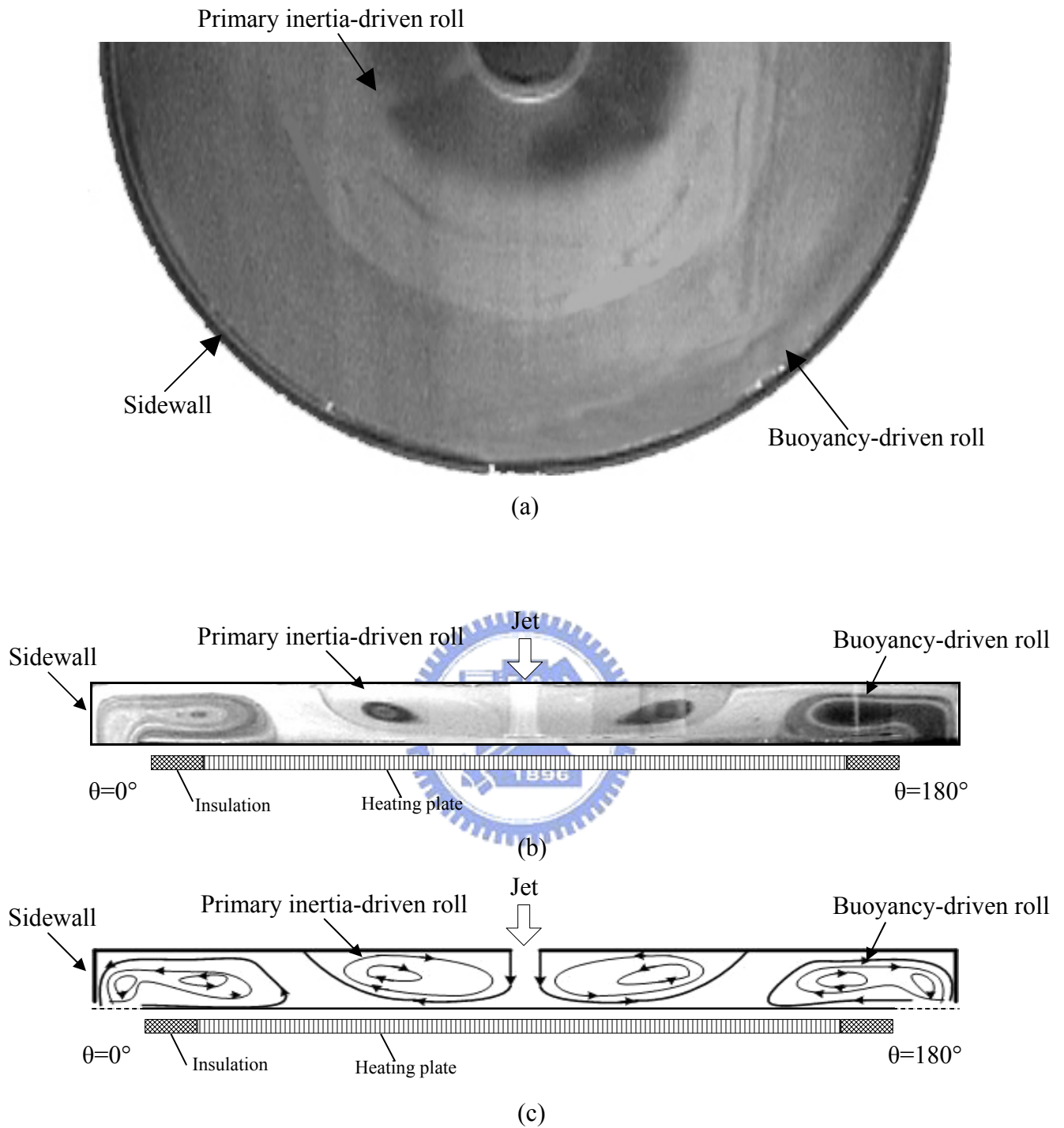
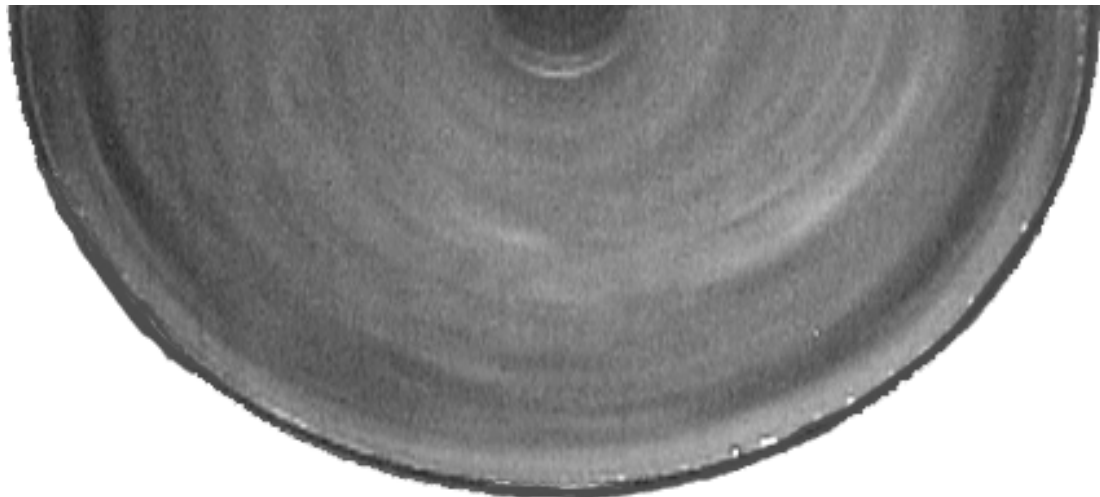
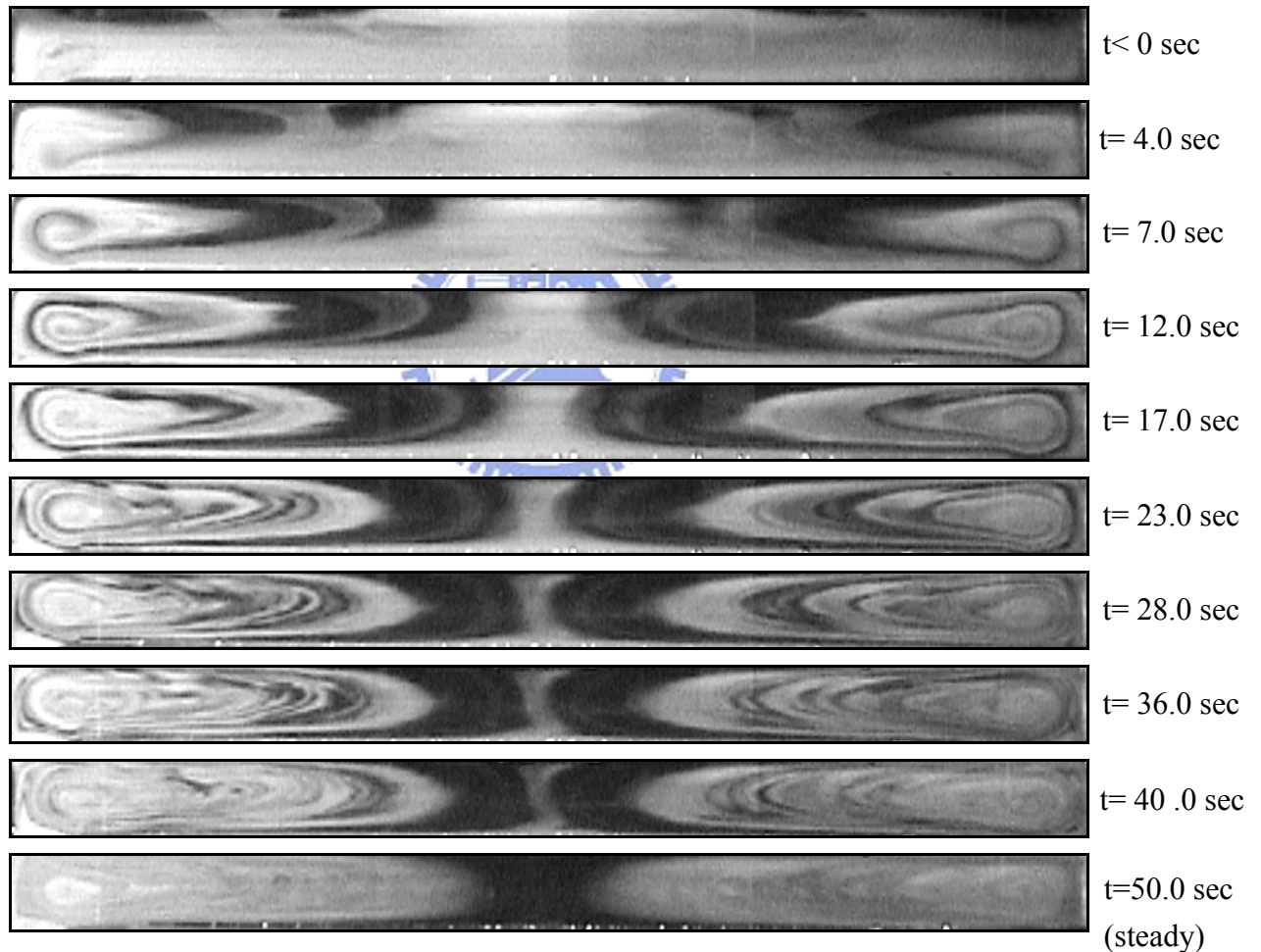


Fig. 4.1 Steady vortex flow pattern in the chamber with $D_j = 10.0$ mm and $H = 20.0$ mm for $Re_j = 270$ ($Q_j = 2.0$ slpm), $Ra = 15,030$ ($\Delta T = 20.0$) and $Re_\Omega = 0$ ($\Omega = 0$ rpm): (a) top view flow photo taken at middle horizontal plane halfway between the pipe exit and heated disk, (b) side view flow photo taken at the vertical plane $\theta = 0^\circ$ & 180° , and (c) the corresponding schematically sketched cross plane vortex flow.



(a) steady state



(b)

Fig. 4.2 Vortex flow formation for $D_j = 10.0$ mm and $H = 20.0$ mm at $Re_j = 0$, $Ra = 0$ and $Re_\Omega = 778$ ($\Omega = 10$ rpm): (a) top view flow photo taken at the middle horizontal plane between the disk and chamber top, and (b) side view flow photo taken at the vertical plane $\theta = 0^\circ$ & 180° .

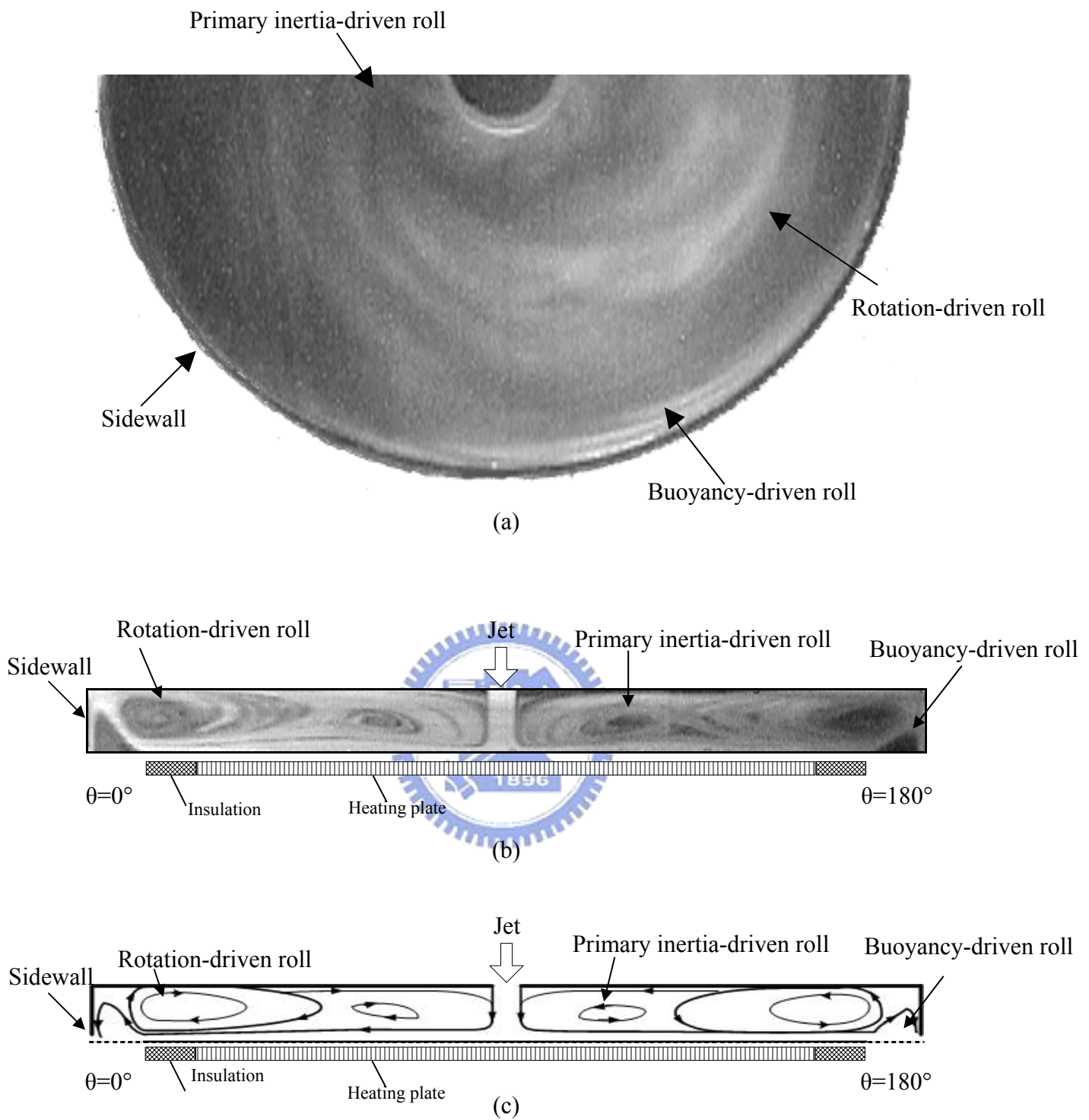


Fig. 4.3 Steady vortex flow pattern for $D_j = 10.0$ mm and $H = 20.0$ mm at $Re_j = 270$ ($Q_j = 2.0$ slpm), $Ra = 15,030$ ($\Delta T = 20.0$) and $Re_\Omega = 2,335$ ($\Omega = 30$ rpm): (a) top view flow photo taken at the middle horizontal plane between the disk and chamber top, (b) side view flow photo taken at the vertical plane $\theta = 0^\circ$ & 180° , and (c) the corresponding schematically sketched cross plane vortex flow.

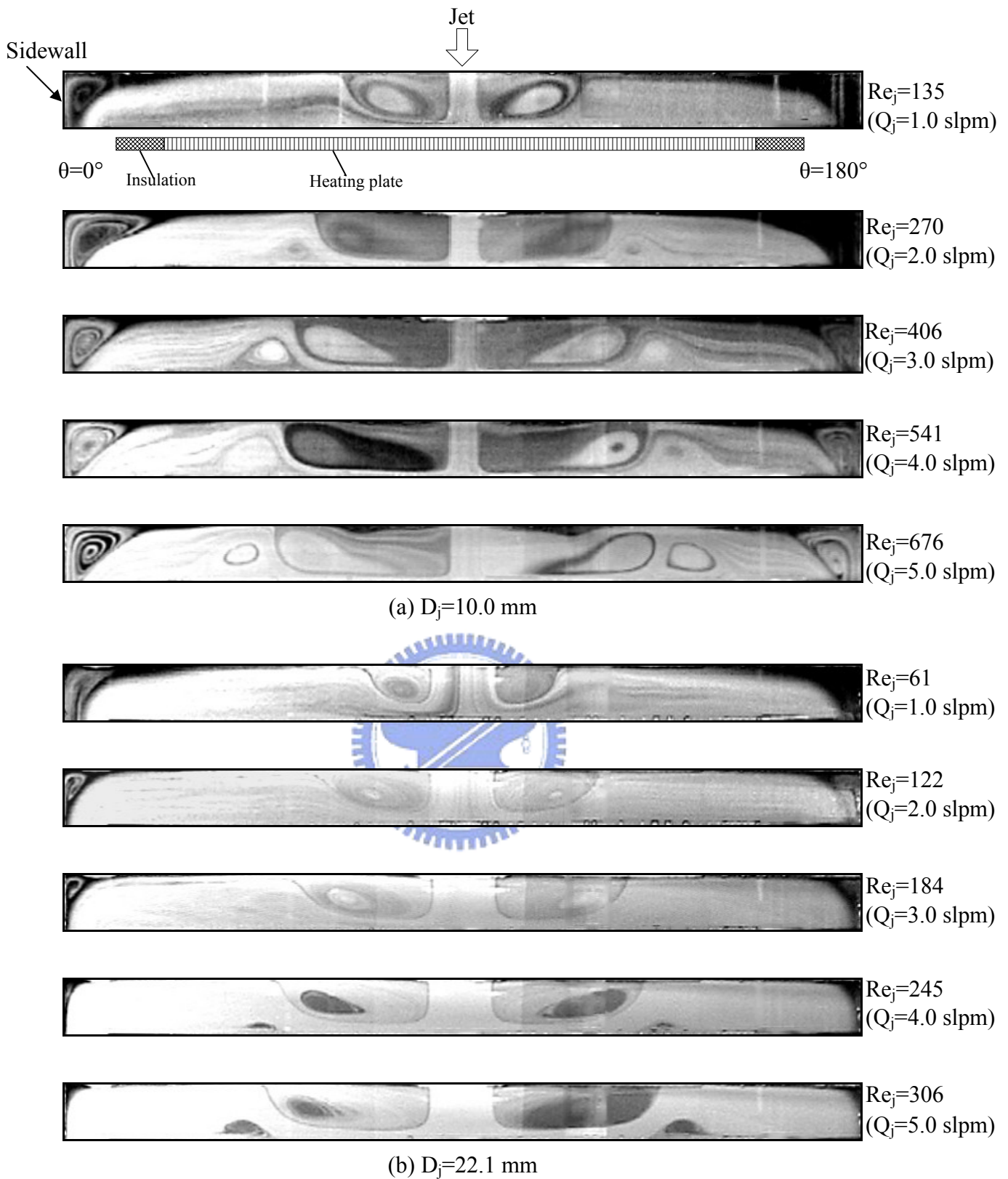


Fig. 4.4 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 0$ ($\Omega = 0$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

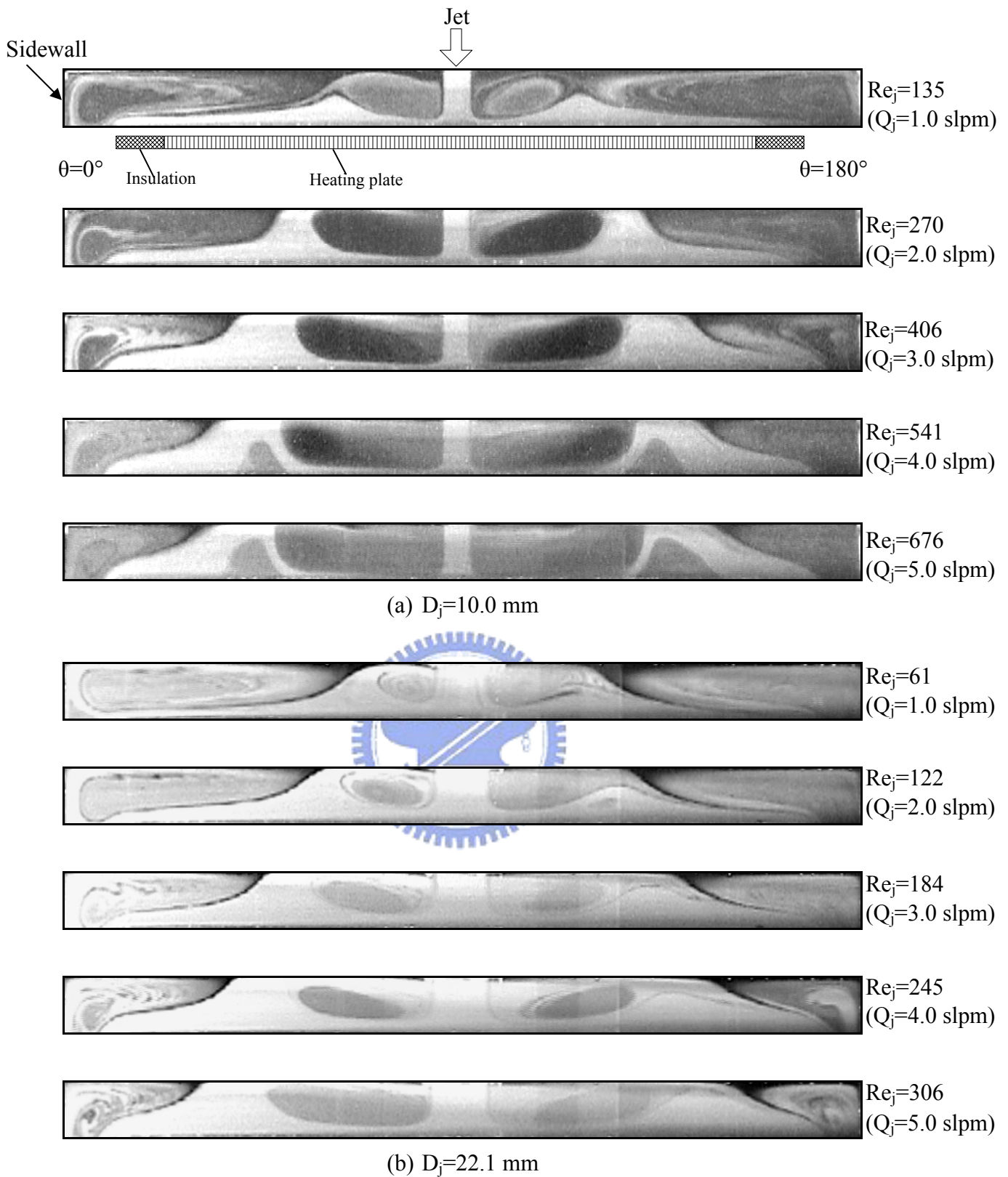


Fig. 4.5 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 778$ ($\Omega = 10$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

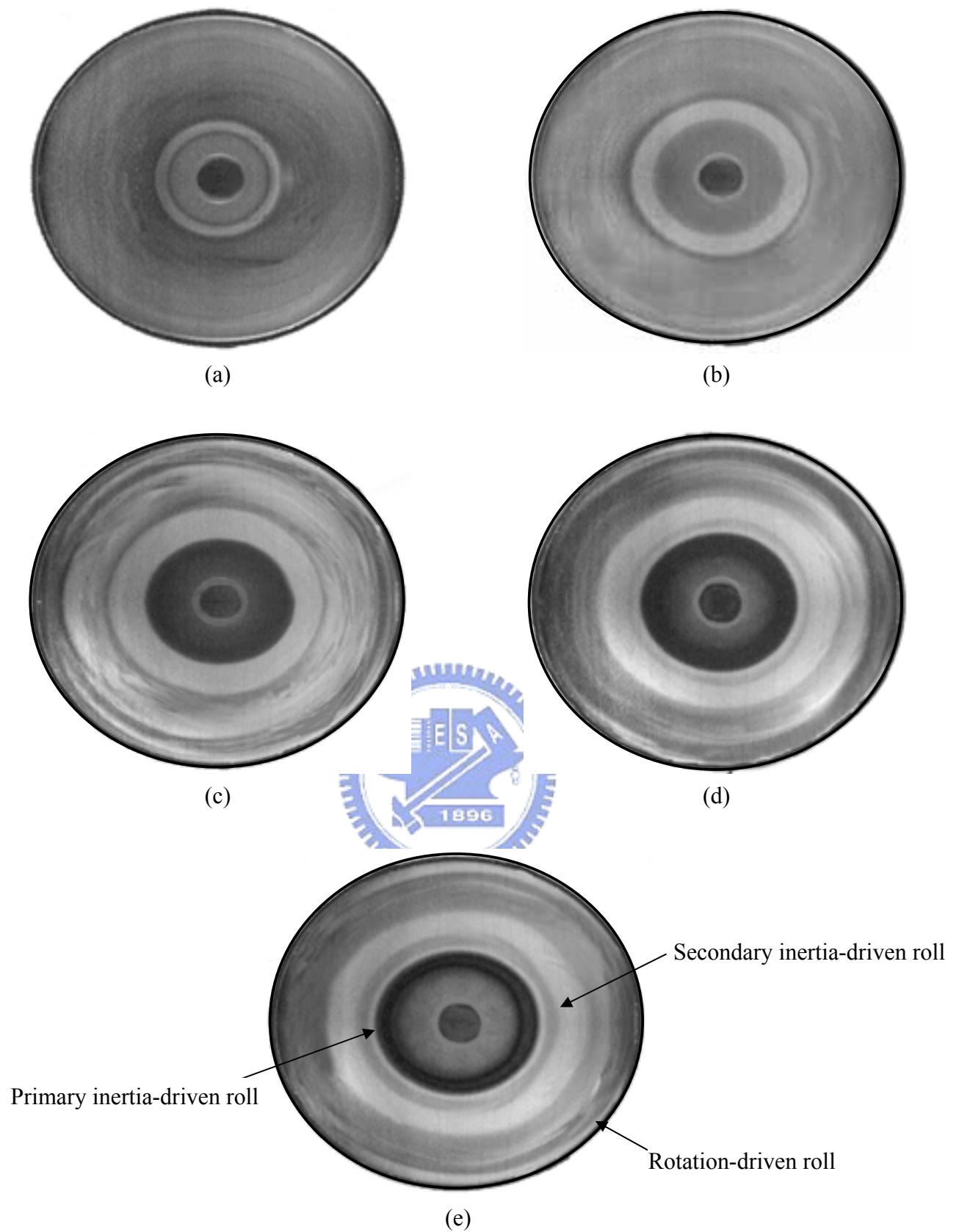


Fig. 4.6 Top view flow photos taken at the middle horizontal plane between the disk and chamber top with $Ra = 0$ ($\Delta T = 0$), $Re_{\Omega} = 778$ ($\Omega = 10$ rpm) and $D_j = 10.0$ mm for $Re_j =$ (a) 135, (b) 270, (c) 406, (d) 541, and (e) 676.

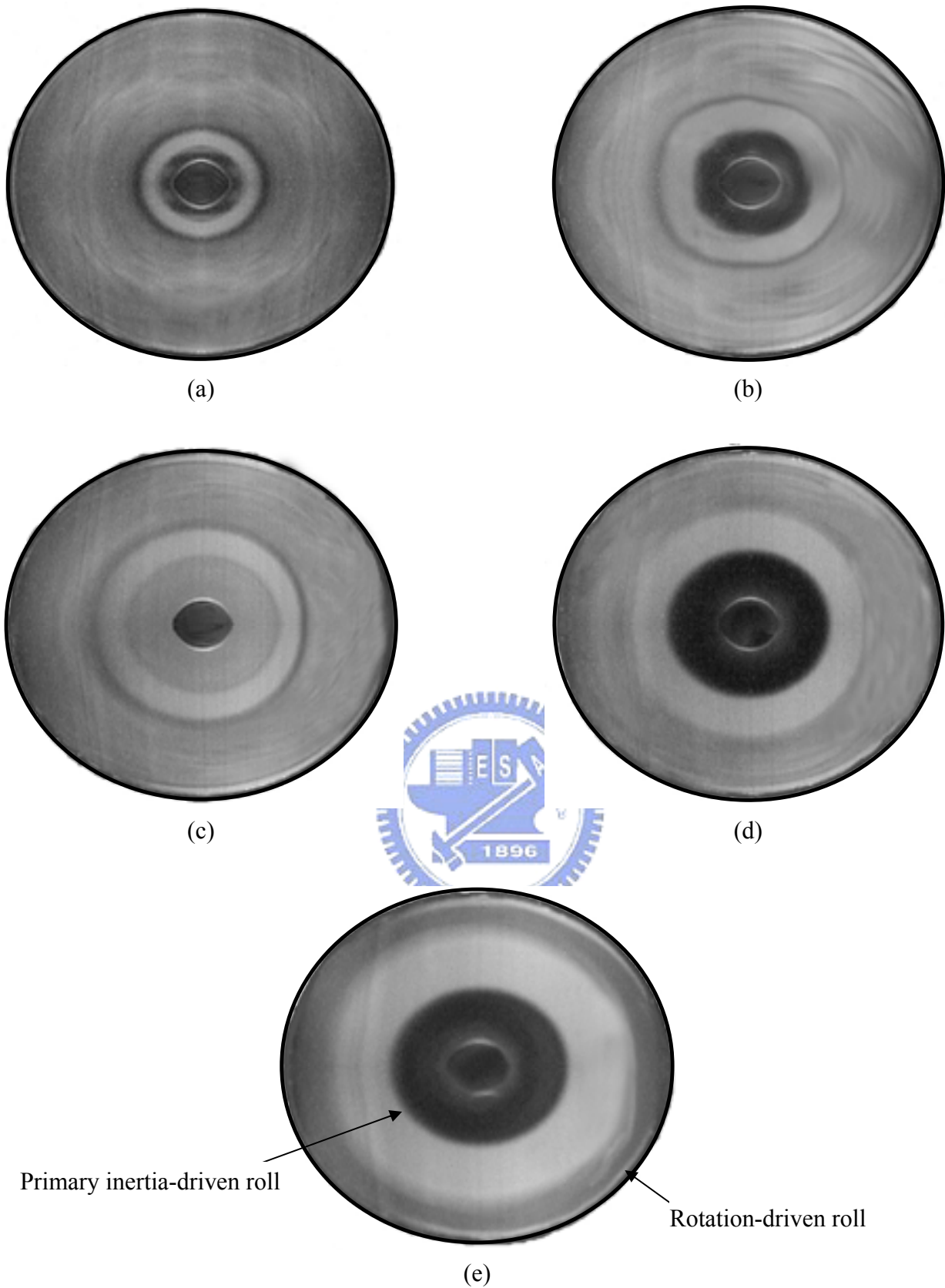


Fig. 4.7 Top view flow photos taken at the middle horizontal plane between the disk and chamber top with $Ra = 0$ ($\Delta T = 0$), $Re_{\Omega} = 778$ ($\Omega = 10$ rpm) and $D_j = 22.1$ mm for $Re_j =$ (a) 61, (b) 122, (c) 184, (d) 245, and (e) 306.

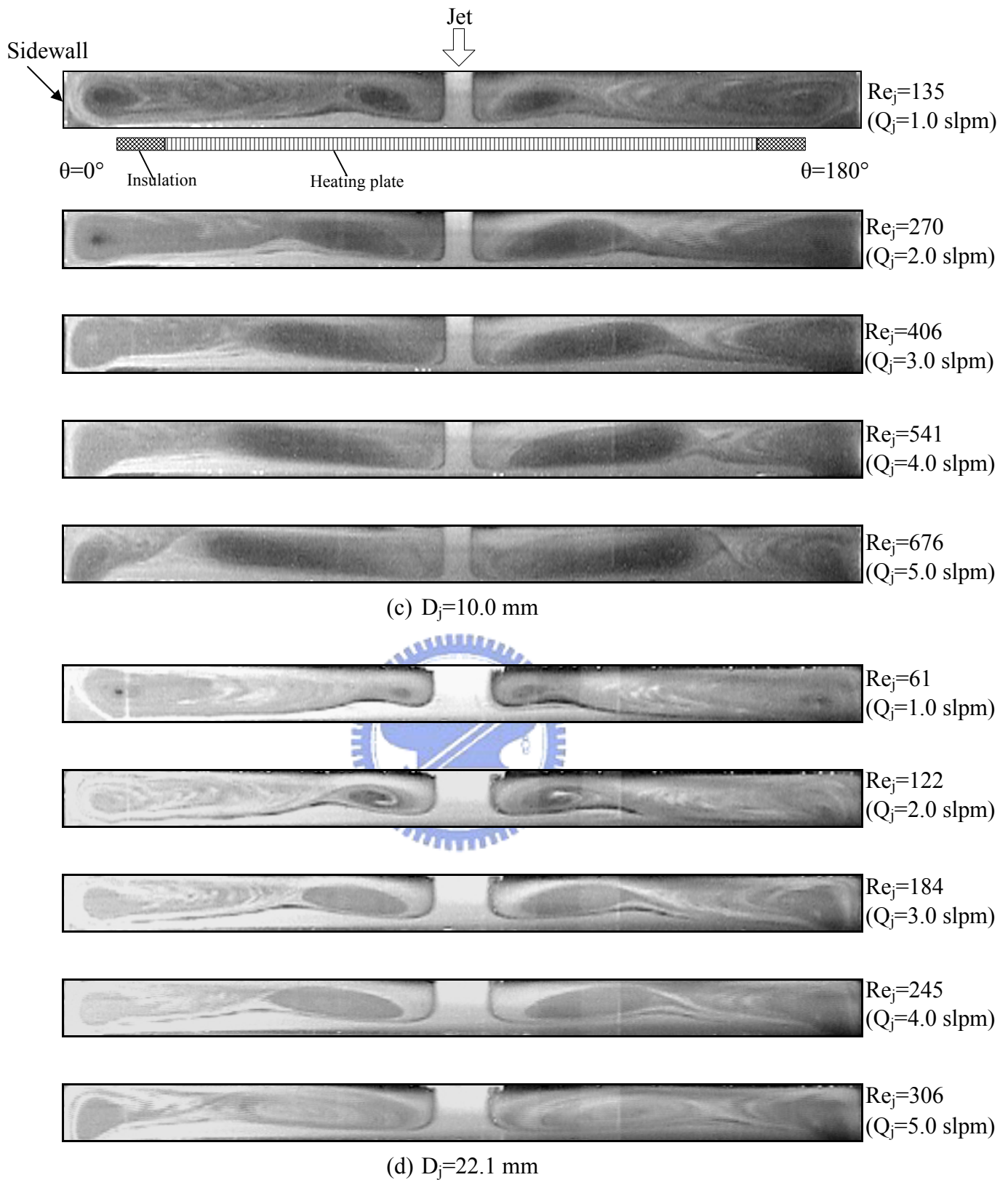


Fig. 4.8 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 1,557$ ($\Omega = 20$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

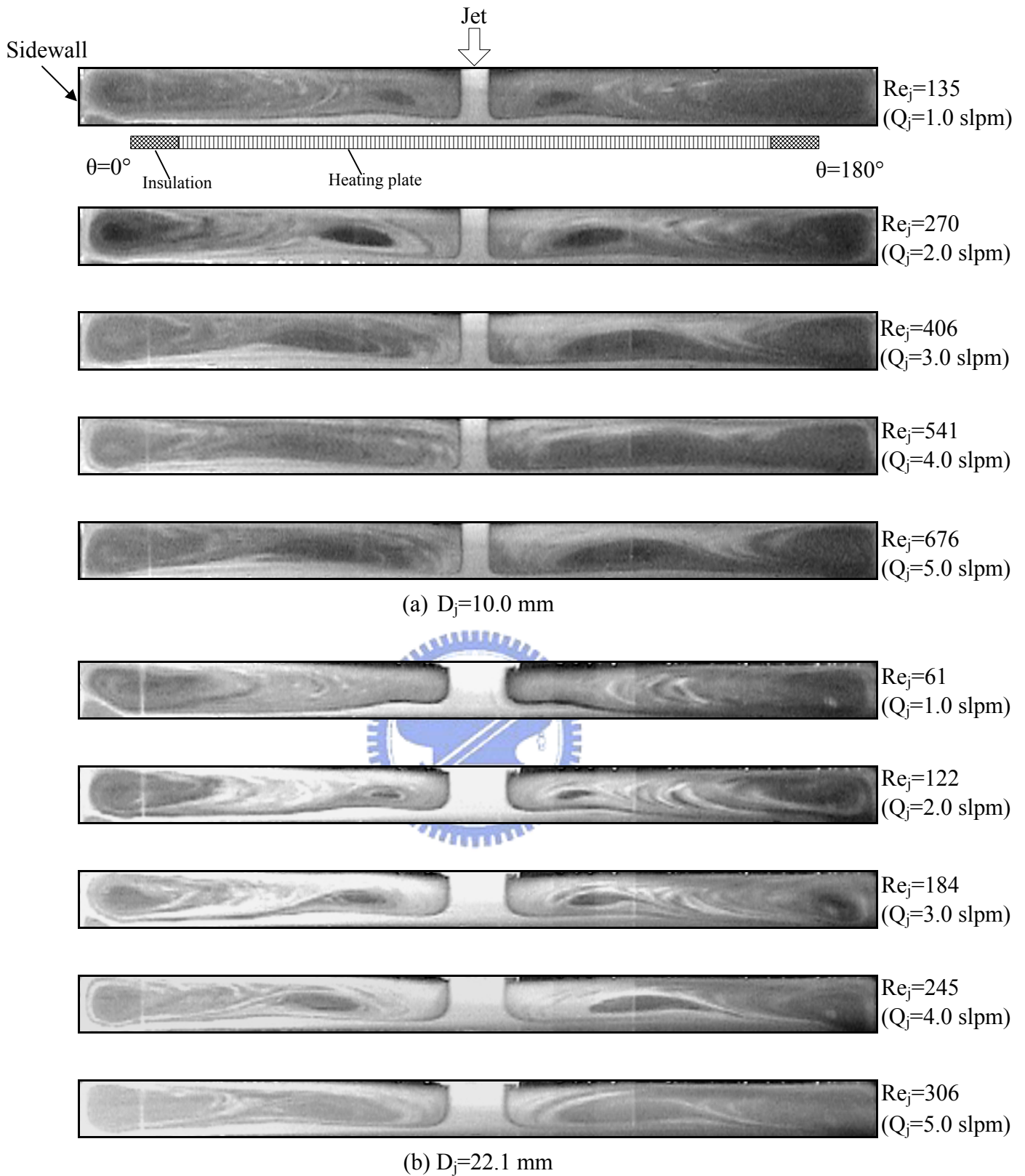


Fig. 4.9 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 2,335$ ($\Omega = 30$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

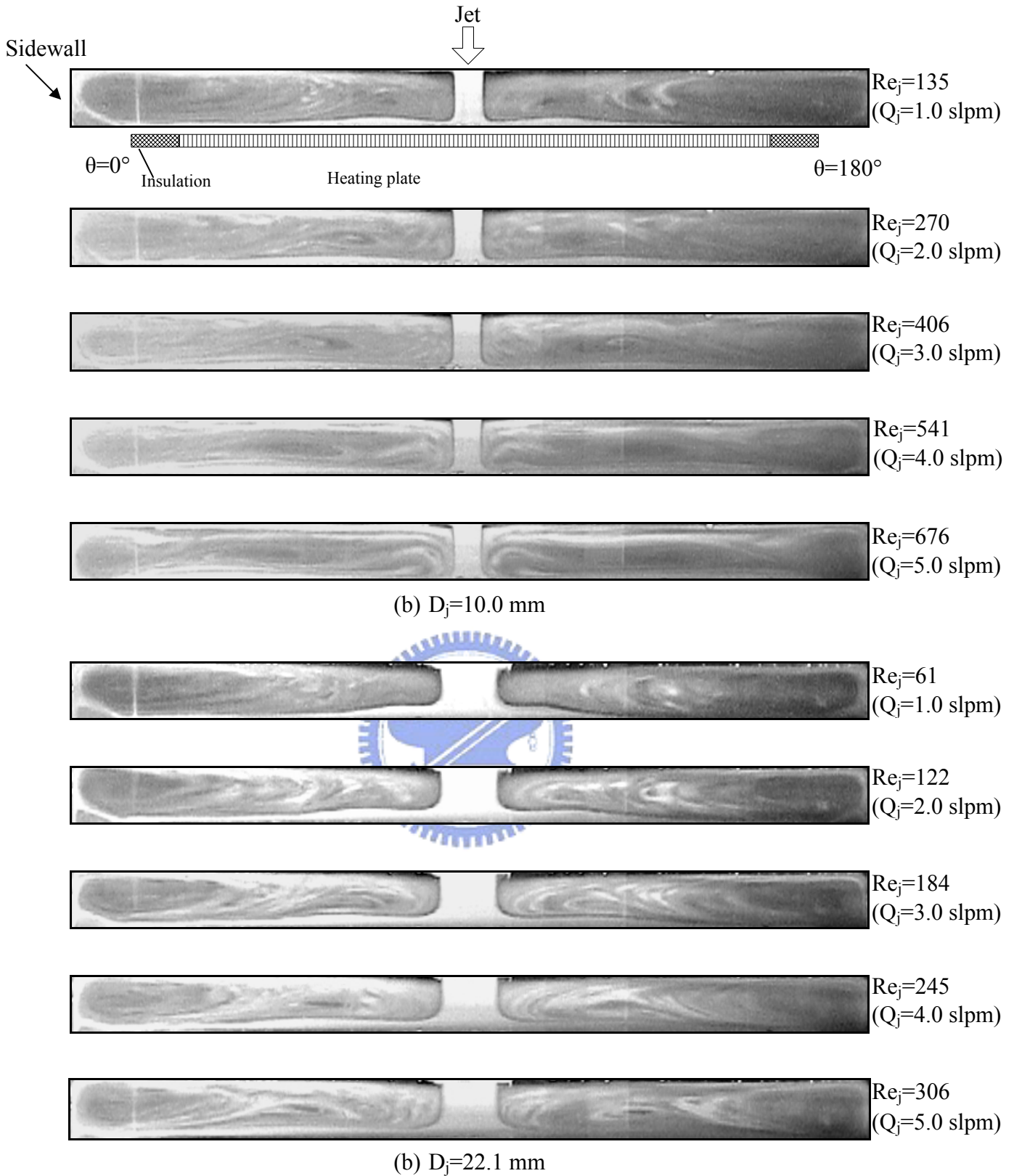


Fig. 4.10 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 3,114$ ($\Omega = 40$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

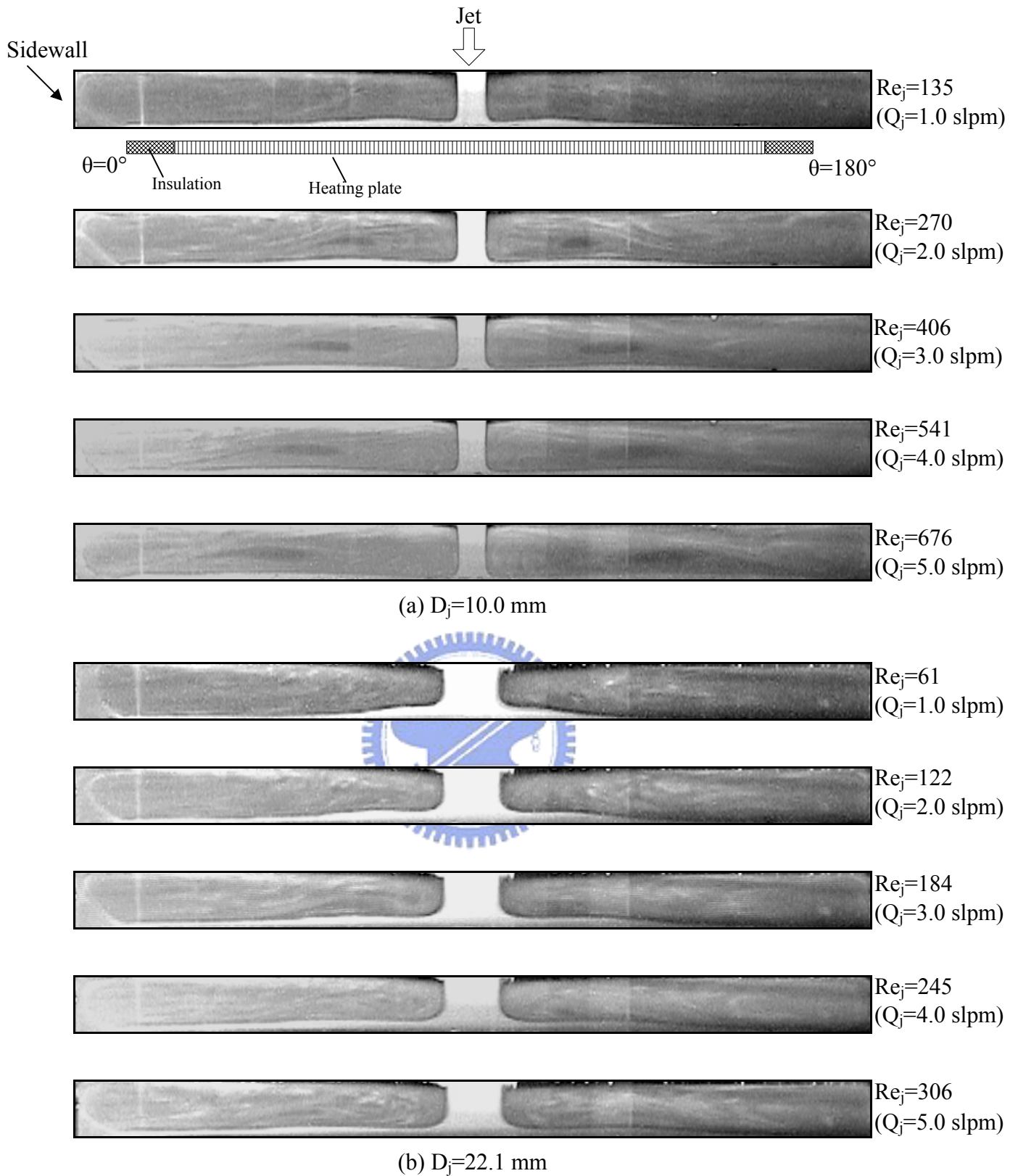


Fig. 4.11 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ & 180° for various jet Reynolds numbers at $Ra = 0$ ($\Delta T = 0$) and $Re_\Omega = 3,892$ ($\Omega = 50$ rpm) for $D_j =$ (a) 10.0 mm and (b) 22.1 mm.

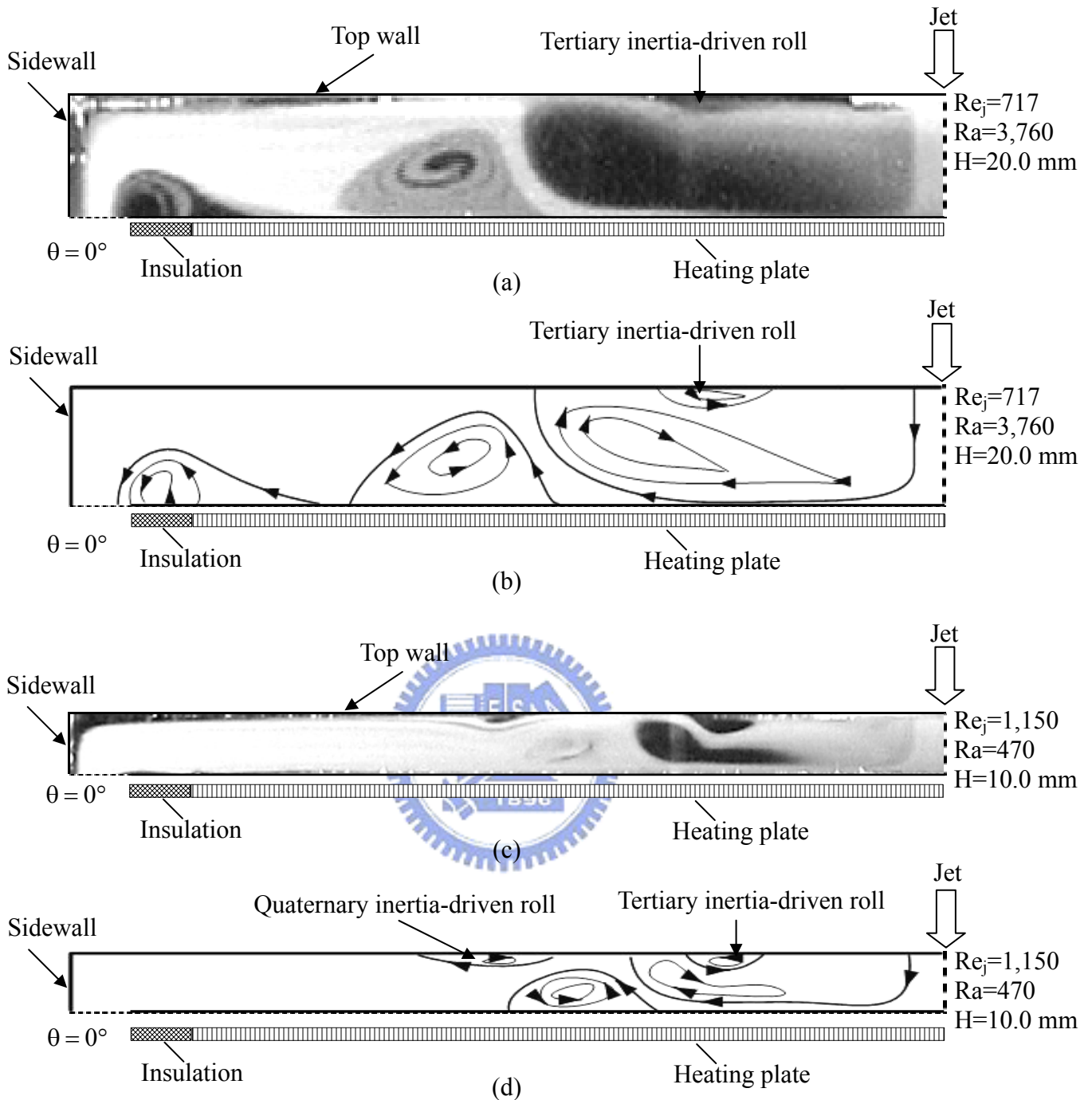


Fig. 4.12 Steady vortex flow pattern in the chamber with $D_j=10.0$ mm: (a) side view flow photo taken at the vertical plane $\theta = 0^\circ$ for $H=20.0$ mm at $Re_j=717$ ($Q_j=5.3$ slpm) and $Ra=3,760$ ($\Delta T = 5.0$) and (b) the corresponding schematically sketched cross plane vortex flow, and (c) side view flow photo for $H=10.0$ mm at $Re_j=1,150$ ($Q_j=8.5$ slpm) and $Ra=470$ ($\Delta T = 5.0$) and (d) the corresponding schematically sketched cross plane vortex flow.

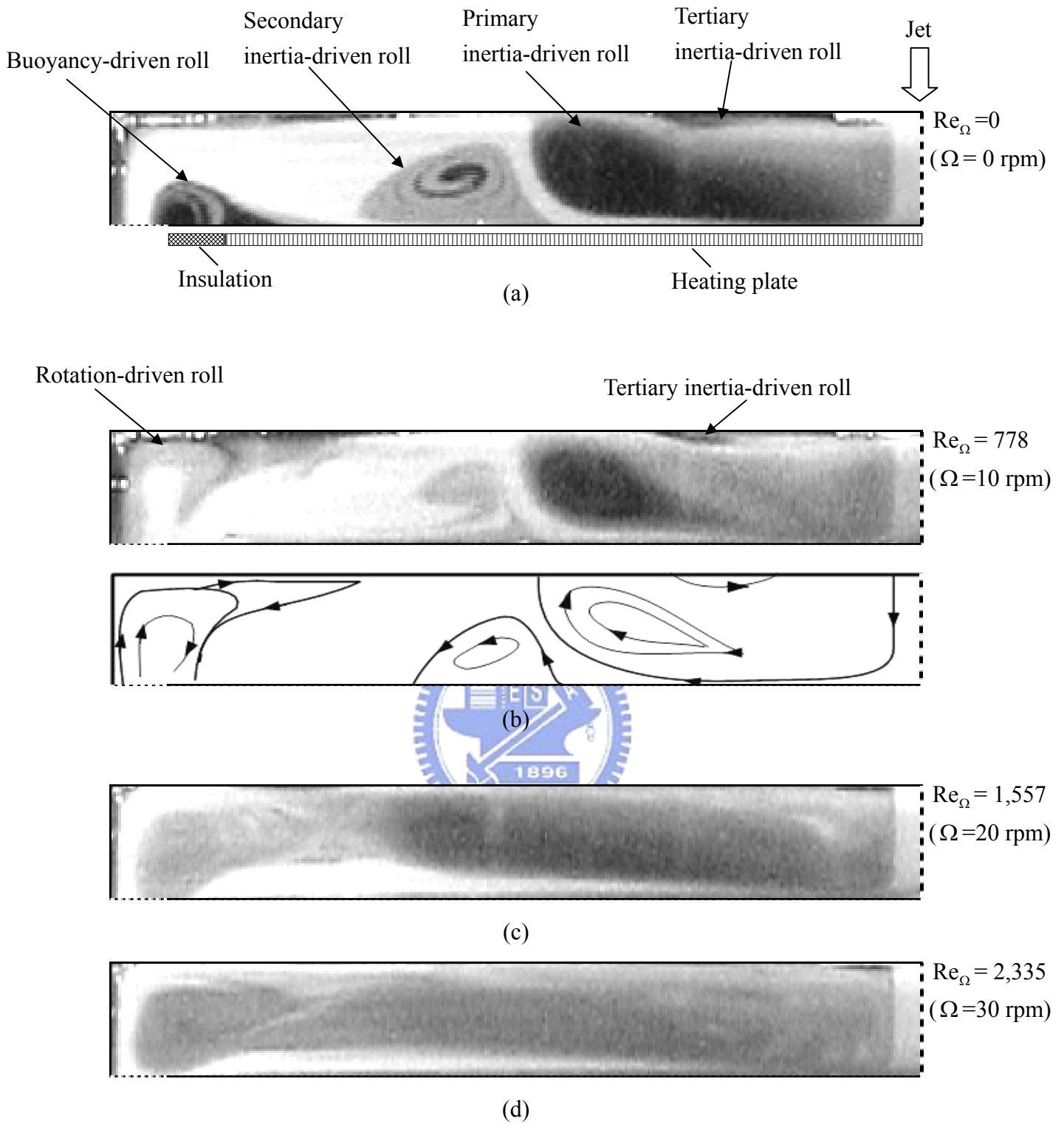
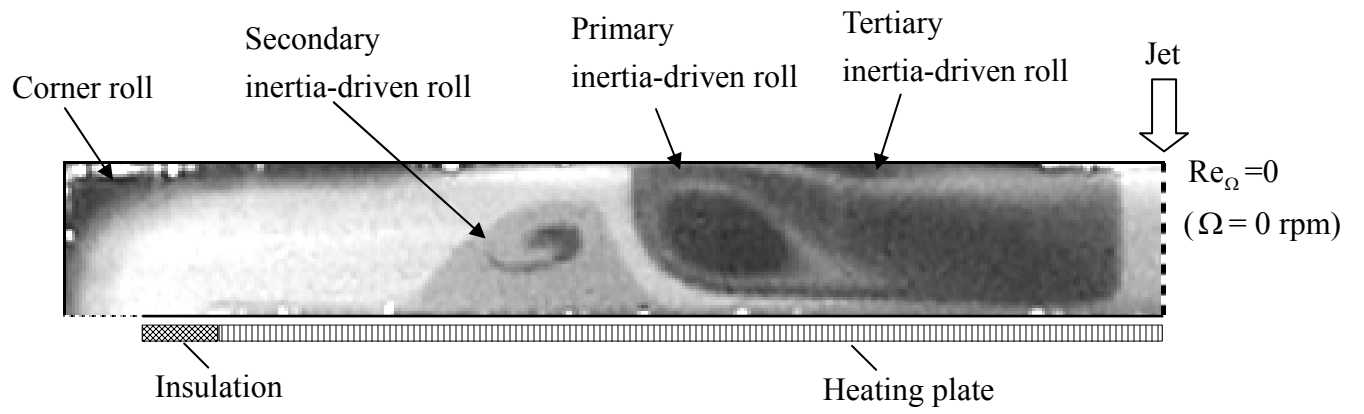
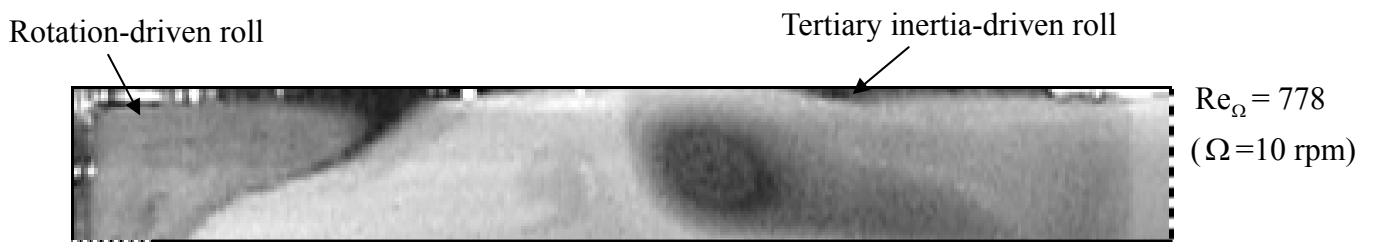


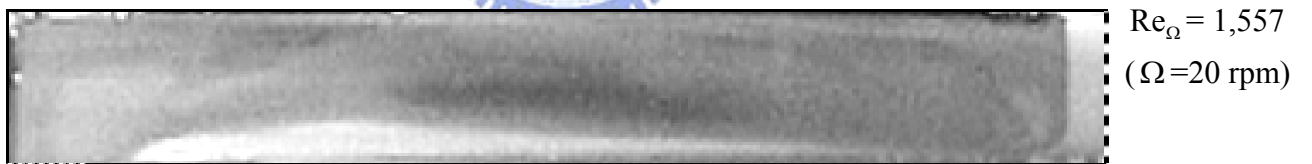
Fig. 4.13 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ at $H=20.0$ mm for $Ra=3,760$ ($\Delta T = 5.0$) and $Re_j = 717$ ($Q_j=5.3$ slpm) for rotational Reynolds number $Re_{\Omega} =$ (a)0, (b)778, (c)1,557, and (d)2,335.



(a)



(b)



(c)



(d)

Fig. 4.14 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ at $H=20.0$ mm for $Ra=0$ ($\Delta T = 0$) and $Re_j = 676$ ($Q_j=5.0$ slpm) for rotational Reynolds number $Re_\Omega =$ (a)0, (b)778, (c)1,557, and (d)2,335.

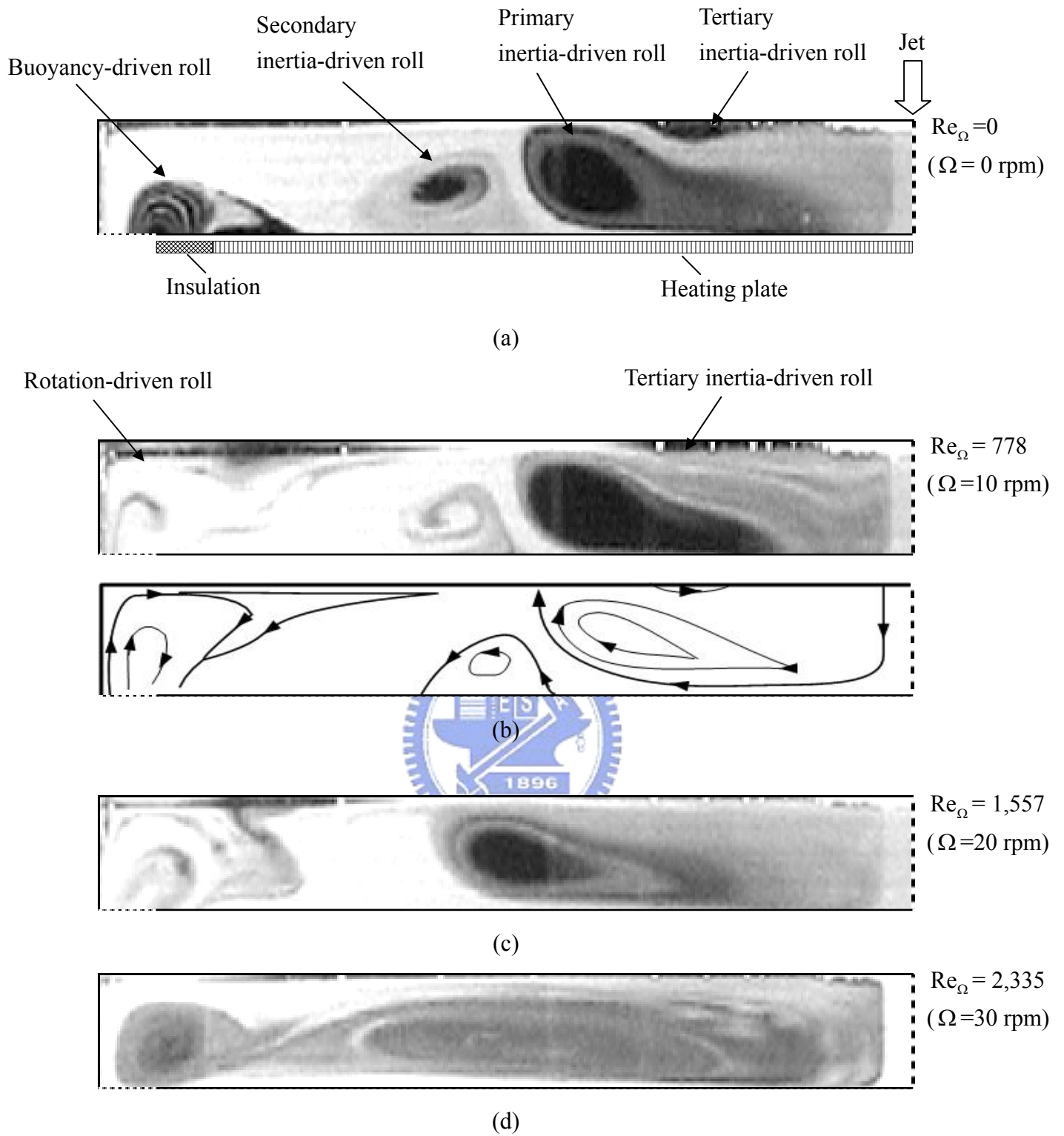
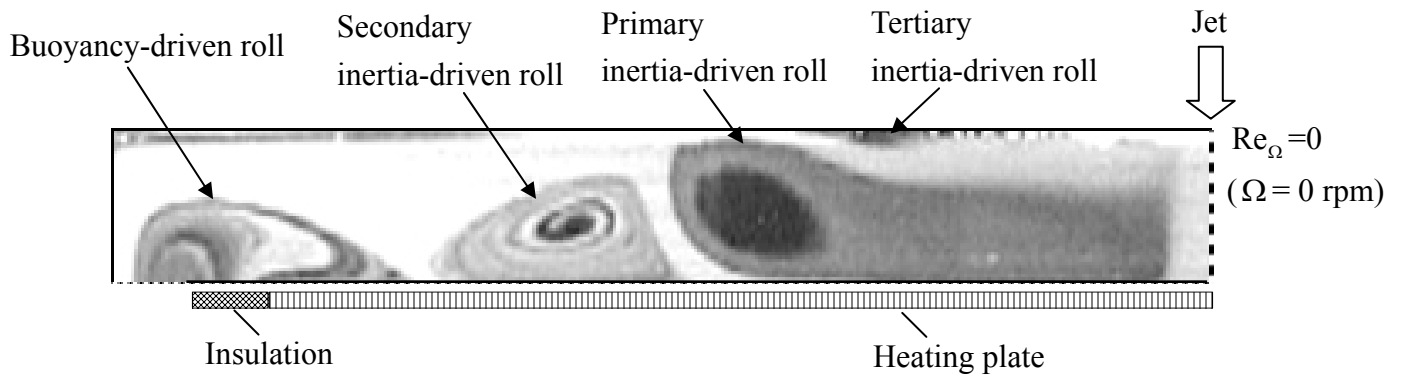
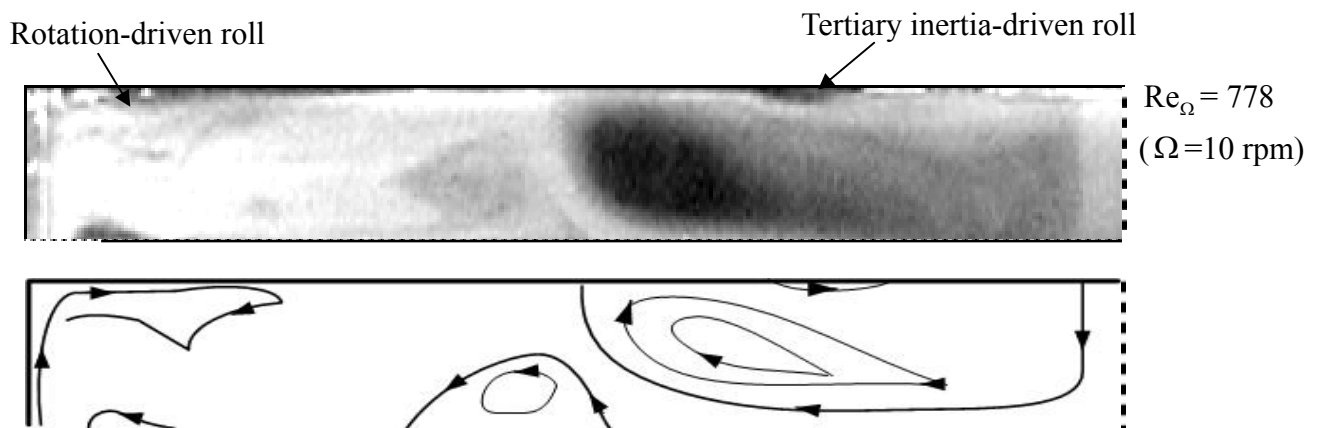


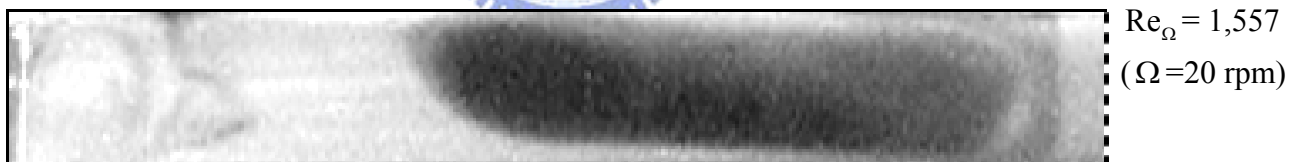
Fig. 4.15 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ at $H=20.0$ mm for $Ra=7,520$ ($\Delta T = 10.0$) and $Re_j = 730$ ($Q_j = 5.4$ slpm) for rotational Reynolds number $Re_{\Omega} =$ (a)0, (b)778, (c)1,557, and (d)2,335.



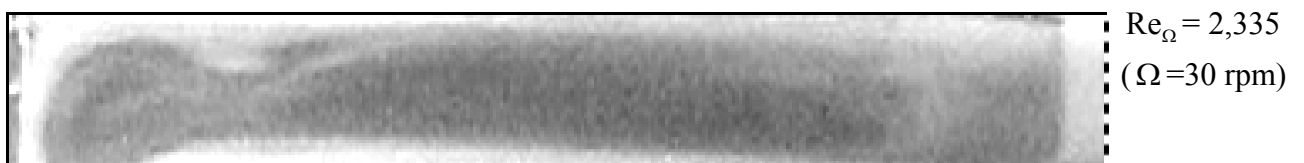
(a)



(b)



(c)



(d)

Fig. 4.16 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ at $H=20.0$ mm for $Ra=11,270$ ($\Delta T=15.0$) and $Re_j=771$ ($Q_j=5.7$ slpm) for rotational Reynolds number $Re_{\Omega} =$ (a)0, (b)778, (c)1,557, and (d)2,335.

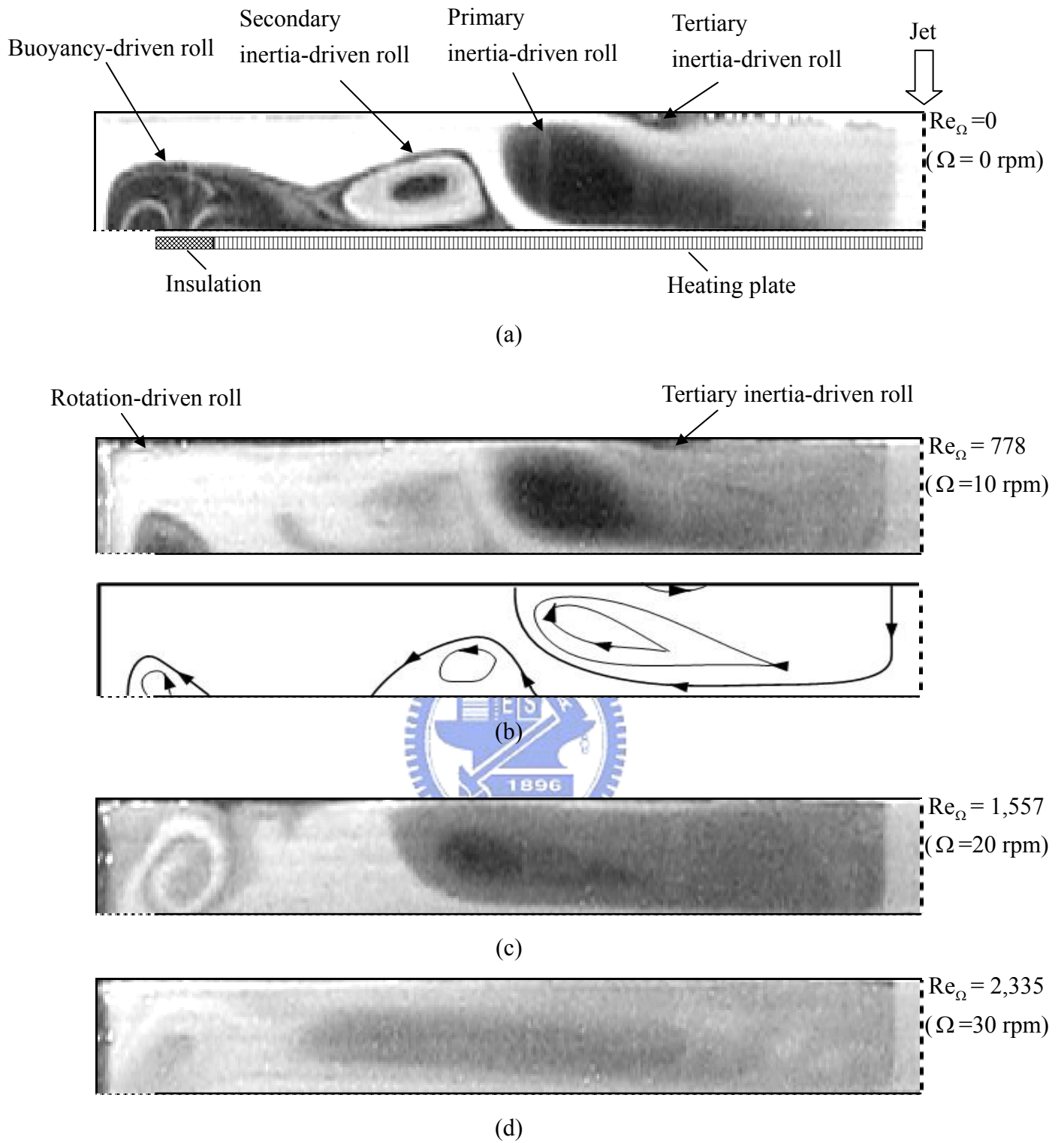


Fig. 4.17 Steady side view flow photos taken at the cross plane $\theta = 0^\circ$ at $H=20.0$ mm for $Ra=15,030$ ($\Delta T=20.0$) and $Re_j= 811$ ($Q_j= 6.0$ slpm) for rotational Reynolds number $Re_{\Omega} =$ (a)0, (b)778, (c)1,557, and (d)2,335.